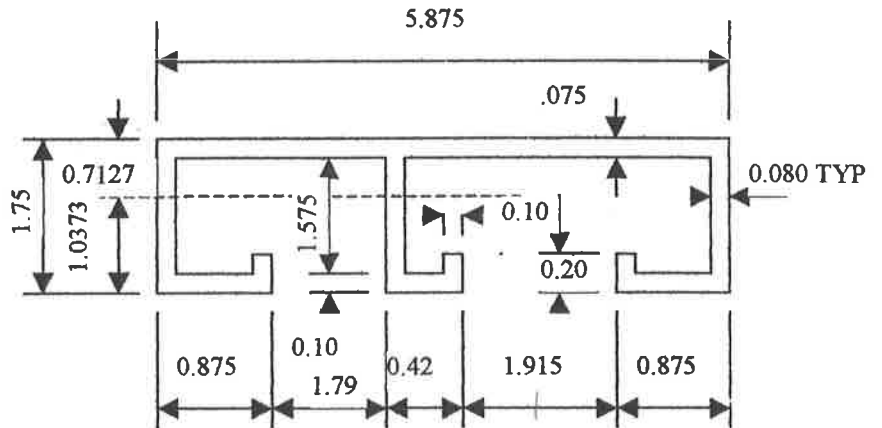


LOAD CALCULATIONS FOR 2x6 ALUMINUM PLANK

$$\begin{aligned}
 A &= 0.075 (5.875) = 0.4406 \text{ in}^2 \\
 A &= 0.10 (0.875)(2) + \\
 &\quad 0.10(0.42) = 0.2170 \text{ in}^2 \\
 A &= 0.08 (1.575)(3) = 0.3780 \text{ in}^2 \\
 A &= 3 (0.10)(0.10) = 0.0300 \text{ in}^2 \\
 A \text{ total} &= 1.0656 \text{ in}^2
 \end{aligned}$$



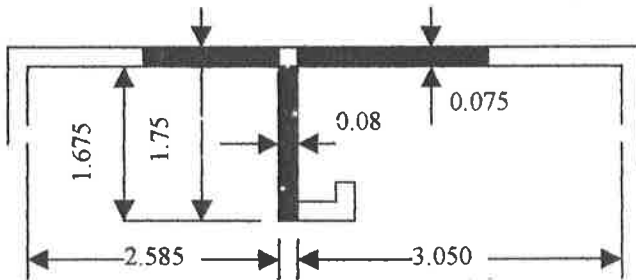
$$\text{N.A.} = \frac{0.4406 (1.7125) + 0.2170 (0.050) + 0.3780 (0.8875) + 0.0300 (0.15)}{1.0656} = 1.0373 \text{ in}$$

$$I = 5.875 (0.075)^3/12 + 0.4406 (0.6752)^2 + 2.17 (0.10)^3/12 + 0.2170(0.9873)^2 +$$

$$3(0.080)(1.575)^3/12 + 0.3780(0.1498)^2 + 3 (0.10)(0.10)^3/12 + 0.03(0.8873)^2 = 0.5231 \text{ in}^4$$

$$S_x \text{ top} = 0.5231 / 0.7127 = 0.7339 \text{ in}^3$$

$$S_x \text{ bottom} = 0.5231 / 1.0373 = 0.5043 \text{ in}^3 \quad r = \sqrt{0.5231 / 1.0656} = 0.7006 \text{ in}$$



Reference: Allowable compressive stress in plank in accordance with Section 3.4.1 of the *Aluminum Design Manual*; The Aluminum Association 900 19th Street N.W. Washington D. C. © 1994

F_{cy} = Compressive Yield Stress = 25 ksi
 B_c = Buckling Formula Constant – Compression
 D_c = Buckling Formula Constant – Compression
 $C_c = 0.41 \times B_c/D_c$ Table 3.3-4

$k_c = 1.12$ Table 3.4-2
 $n_y = 1.65$ Table 3.4-1
 $n_u = 1.95$ Table 3.4-1

$$B_c = F_{cy} (1 + \sqrt{F_{cy}/2250}) \text{ Table 3.3-4}$$

$$B_c = 25 (1 + \sqrt{25/2250}) = 27.64 \text{ ksi}$$

$$D_c = B_c/10 \sqrt{B_c/E} \text{ Table 3.3-4}$$

$$D_c = 27.64/10 \sqrt{27.64/10100} = 0.1445$$

$$C_c = 0.41 (27.64/0.1445) = 78.42$$

$$F_c = 25 \text{ ksi for 6063-T6 Table 3.3-1}$$

$$S1 = [Bc - (nu Fcy) / (kc ny)] / Dc \quad \text{Eq. 3.4.9.2-4}$$

$$S1 = [27.64 - (1.95 \times 25 / 1.12 \times 1.65)] / 0.1445 = 8.72$$

$$S2 = Cc = 78.42 \quad \text{Eq. 4.4.9.2-5} \quad b = (2.585 + 3.050) / 2 = 2.8175$$

$$Fc = Fcy / kc ny = 25 / 1.12 \times 1.65 = 13.53 \quad \text{If } \lambda < S1 \quad \text{Eq. 3.4.9.2-1}$$

$$A_{\text{stem}} = 1.675 (0.080) = 0.1340 \text{ in}^2$$

$$A_{\text{flange}} = 2.8175 (0.075) = 0.2113 \text{ in}^2$$

$$\text{Area total} = 0.3453 \text{ in}^2$$

$$N.A. = \frac{0.080 (1.675)(0.8375) + 2.8175 (0.075)(1.7125)}{0.3453} = 1.373 \text{ in}$$

$$I_o = 0.080 (1.675)^3 / 12 + 0.134(0.5351)^2 + 2.818 (0.075)^3 / 12 + 0.2113 (0.3399)^2$$

$$I_o = 0.0942 \text{ in}^4$$

$$\lambda = 4.62(b/t) \sqrt{\frac{1 + A_s/bt}{1 + \sqrt{1 + (10.67 I_o) / (bt)}}}} = 4.62(2.8175/0.075) \sqrt{\frac{1 + 0.134/(2.8175)(0.075)}{1 + \sqrt{1 + \frac{10.67(0.0942)}{2.8175(0.075)^3}}}}$$

$$\lambda = 40.442$$

$$Fc = (1/nu)(Bc - Dc\lambda) = (1/1.95)[27.64 - 0.1445(40.442)] \quad \text{Eq. 3.4.9.2-2}$$

$$Fc = 11.18 \text{ ksi} \quad \text{If } \lambda \text{ is between } S1 \text{ and } S2 \quad \leftarrow \text{ [Use this one]}$$

$$Fc = \pi^2 E / nu \lambda^2 = \pi^2 (10100) / 1.95 (40.442)^2 = 31.26 \text{ ksi} \quad \text{If } \lambda \text{ is } > S2 \quad \text{Eq. 3.4.9.2-3}$$

$$Fty = 25 \text{ ksi} \quad \text{Table 3.3-1}$$

Foot and seat planks must be capable of supporting 120 plf. NRS uses a 2 x 10 (9 1/2") as our narrowest seat or foot board. Review the 2 x 6 plank based on the same area load as a 120 plf on a 9 1/2" plank.

$$\text{Load per inch of width} = 120 / 9.5 = 12.63 \text{ lbs. per in}$$

$$12.63 (5.875) = 74.20 \text{ plf}$$

Maximum frame spacing 6'-0"

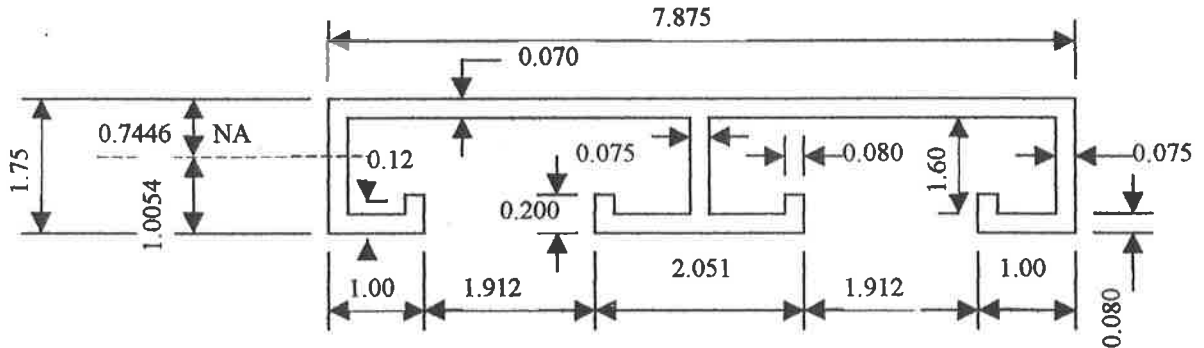
$$M = 74.20 \times 6^2 / 8 = 333.91 \text{ ft-lbs.}$$

$$\text{fb compression} = 333.91 \times 12 / 0.7339 = 5460 \text{ psi} < 11.18 \text{ ksi}$$

$$\text{fb tension} = 333.91 \times 12 / 0.5043 = 7946 \text{ psi} < 25 \text{ ksi}$$

The 2 x 6 plank is adequate for National Recreation Systems' applications.

LOAD CALCULATIONS FOR MEDIUM DUTY 2 X 8 ALUMINUM PLANK



$$\begin{aligned}
 \text{Area} &= 7.875 \times 0.070 = 0.5513 \text{ in}^2 \\
 &= (2 \times 1.00 + 2.051) \times 0.080 = 0.3241 \text{ in}^2 \\
 &= 1.600 \times 0.075 \times 3 = 0.3600 \text{ in}^2 \\
 &= 4 \times 0.080 \times 0.120 = 0.0384 \text{ in}^2 \\
 \text{Total Area} &= 1.2738 \text{ in}^2
 \end{aligned}$$

$$\text{N.A.} = \frac{0.5513 \times 1.715 + 0.3241 \times 0.04 + 0.36 \times 0.88 + 0.0384 \times 0.14}{1.2738} = 1.0054 \text{ in}$$

$$\begin{aligned}
 I &= 7.875(0.070^3)/12 + 0.5513(0.7096^2) + 4.051 (0.080^3)/12 + 0.3241 (0.9654^2) + \\
 &3 (0.075)(1.6^3)/12 + 0.36 (0.1254)^2 + 4 (0.08)(0.12^3)/12 + 0.0384 (0.8654^2) = \\
 I &= 0.6913 \text{ in}^4
 \end{aligned}$$

$$S_x \text{ top} = 0.6913/0.7446 = 0.9284 \text{ in}^3$$

$$S_x \text{ bottom} = 0.6913 / 1.0054 = 0.6876 \text{ in}^3$$

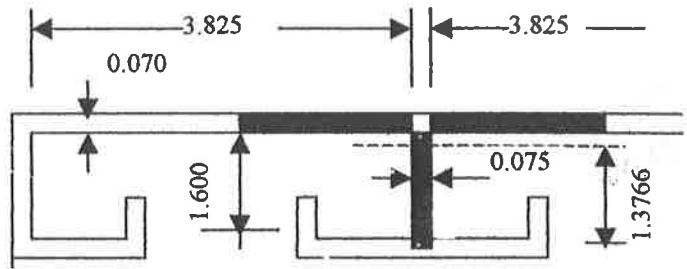
$$r_x = \sqrt{0.6913 / 1.2738} = 0.7367 \text{ in}$$

Reference: Allowable compressive stress in plank in accordance with Section 3.4.9.2 and allowable tension stress in accordance with Section 3.4.1 of the Aluminum Design Manual; The Aluminum Association 900 19th Street N.W. Washington D.C. 20006 © 1994.

F_{cy} = Compressive Yield Strength = 25 ksi** n_y = 1.65 Table 3.4-1
 B_c = Buckling Formula Constant – Compression n_u = 1.95 Table 3.4-1
 D_c = Buckling Formula Constant – Compression k_c = 1.12 Table 3.4-2
 C_c = $0.41 \times B_c/D_c$ Table 3.3-4

** F_{cy} = 25 ksi and F_{ty} = 25 ksi for 6063-T6 Table 3.3-1

$$B_c = F_{cy} \left(1 + \sqrt{F_{cy} / 2250} \right) \quad D_c = B_c / 10 \sqrt{B_c / E} \rightarrow \text{Table 3.3-4}$$



$$B_c = 25 \left(1 + \sqrt{\frac{25}{2250}} \right)$$

$$B_c = 27.64$$

$$D_c = \frac{27.64}{10} \sqrt{\frac{27.64}{10100}} = 0.1445$$

$$C_c = 0.41 (27.64) / 0.1445 = 78.42$$

$$S_1 = [B_c - (n_u \times F_{cy}) / (k_c \times n_y)] / D_c \quad \text{Eq. 3.4.9.2-4}$$

$$S_1 = [27.64 - (1.95 \times 25) / (1.12 \times 1.65)] / 0.1445 = 8.72$$

$$S_2 = C_c = 78.42 \quad \text{Eq. 3.4.9.2-5}$$

$$A_{\text{stem}} = 1.600 (0.075) = 0.12000 \text{ in}^2$$

$$A_{\text{flange}} = 3.825 (0.07) = 0.26775 \text{ in}^2$$

$$A_{\text{total}} = 0.38775 \text{ in}^2$$

$$N.A. = \frac{0.26775 (1.6350) + 0.12 (0.80)}{0.38775} = 1.3766 \text{ in}$$

$$I_o = 0.075 (1.600)^3 / 12 + 0.12 (0.5766^2) + 3.825 (0.07)^3 / 12 + 0.26775 (0.2584)^2 = 0.08348 \text{ in}^4$$

$$\lambda = 4.62 (b/t) \sqrt{\frac{1 + A_s/bt}{1 + \sqrt{1 + 10.67 I_o / bt^3}}} \quad \text{Eq. 3.4.9.2-6}$$

$$\lambda = 4.62 (3.825/0.070) \sqrt{\frac{1 + 0.1200/0.070 (3.825)}{1 + \sqrt{1 + 10.67 (0.08348)/3.825 (0.07)^3}}} = 58.385$$

$$F_c = F_{cy} / k_c n_y = 25 / 1.12 \times 1.65 = 13.52 \text{ ksi} \quad \text{Eq. 3.4.9.2-1}$$

If $\lambda < S_1$

$$F_c = 1/n_u [B_c - D_c \lambda] = 1/1.95 [27.64 - 0.1445(58.385)] = 9.847 \quad \text{Eq. 3.4.9.2-2}$$

If λ is between S_1 and S_2 .

Use This One

$$F_c = \pi^2 E / n_u \lambda^2 = \pi^2 10100 / 1.95 (58.385)^2 = 14.996 \text{ ksi} \quad \text{Eq. 3.4.9.2-3}$$

If $\lambda > S_2$

Foot and seat planks must be capable of supporting 120 plf. NRS uses a 2x10 (9 1/2") as our narrowest seat or foot board. Review the 2x8 plank based on the same area load.

$$\text{Load per inch of width} = 120 / 9.5 = 12.63 \text{ lbs/inch}$$

$$12.63 (7.875) = 99.5 \text{ plf}$$

$$\text{Maximum frame spacing} = 6' - 0''$$

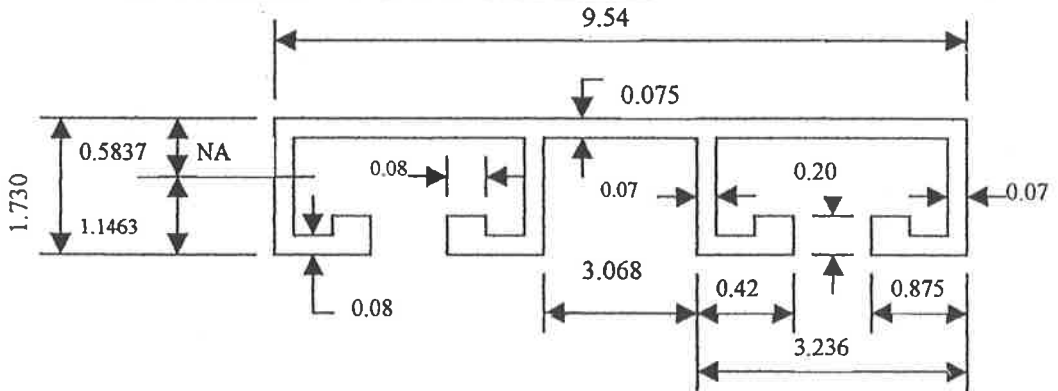
$$M = 99.5 (6^2) / 8 = 448 \text{ ft-lbs}$$

$$\text{fb tension} = 448 (12) / 0.6876 = 7818 \text{ psi} < 25 \text{ ksi}$$

$$\text{fb compression} = 448(12) / 0.9284 = 5791 \text{ psi} < 9.847 \text{ ksi}$$

The 2x8 plank is adequate for National Recreation Systems' applications.

LOAD CALCULATIONS FOR MEDIUM DUTY 2 x 10 ALUMINUM PLANK



Area = 9.540 x 0.075 =	0.7155 in ²
Area = [2(0.875) + 2(0.420)](0.080) =	0.2072 in ²
Area = 4(1.575)(0.070) =	0.4410 in ²
Area = 4(0.080)(0.120) =	0.0384 in ²
Total Area =	1.4021 in²

$$NA = \frac{0.7155(1.6925) + 0.2072(0.04) + 0.441(0.8675) + 0.0384(0.140)}{1.4021} = 1.1463 \text{ in}$$

$$I = 9.540(0.075)^3/12 + 0.7155(0.5462)^2 + 2.59(0.080)^3/12 + 0.2072(1.1063)^2 + 4(0.070)(1.575)^3/12 + 0.441(0.2788)^2 + 4(0.080)(0.12)^3/12 + 0.0384(1.0063)^2 = 0.6319 \text{ in}^4$$

$$S_x\text{-top} = 0.6319/0.5837 = 1.0826 \text{ in}^3$$

$$S_x\text{-bottom} = 0.6319/1.1463 = 0.5513 \text{ in}^3$$

$$r_x = \sqrt{0.6319/1.4021} = 0.6713 \text{ in}$$

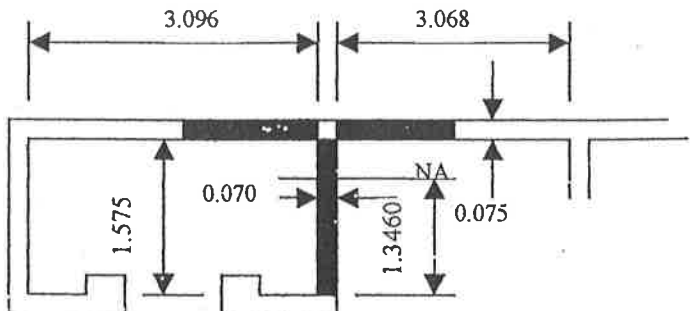
Reference: Allowable compressive stress in plank in accordance with Section 3.4.9.2 and allowable tension stress in accordance with Section 3.4.1 of the *Aluminum Design Manual*; The Aluminum Association 900 19th Street, N.W. Washington D.C. 20006 © 1994.

F_{cy} = Compressive Yield Strength = 25 ksi**
 B_c = Buckling Formula Constant - Compression
 D_c = Buckling Formula Constant - Compression
 C_c = 0.41 x B_c/D_c Table 3.3-4

$$B_c = F_{cy} \{ 1 + (F_{cy}/2250) \}^{1/2} \text{ Table 3.3-4}$$

$$B_c = 25 \{ 1 + (25/2250) \}^{1/2} = 27.64 \text{ ksi}$$

$$D_c = (B_c/10) \{ (B_c/E) \}^{1/2} \text{ Table 3.3-4}$$



$$D_c = 27.64/10 [(27.64/10100)^{1/2}] = 0.1445 \quad b = \frac{3.096 + 3.068}{2} = 3.082$$

$$C_c = 0.41 (27.64/0.1445) = 78.42$$

** Fc=25 ksi for 6063-T6 Table 3.3-1

$$S_1 = [B_c - (n_u F_{cy}/k_c n_y)] / D_c = [27.64 - (1.95 \times 25 / 1.12 \times 1.65)] / 0.1445 = 8.719 \quad \text{Eq. 3.4.9.2-4}$$

$$S_2 = C_c = 78.42 \quad \text{Eq. 3.4.9.2-5}$$

$$F_c = F_{cy} / k_c n_y = 25 / (1.12)(1.65) = 13.53 \quad \text{Eq. 3.4.9.2-1}$$

$$\text{Area stem} = 1.575(0.070) = 0.1103 \text{ in}^2$$

$$\text{Area flange} = \frac{3.082(0.075)}{0.3415} = 0.2312 \text{ in}^2$$

$$\text{Area Total} = 0.3415 \text{ in}^2$$

$$N_A = \frac{0.1103(0.7875) + 0.2312(1.6125)}{0.3415} = 1.3460 \text{ in}$$

$$I_o = 0.070(1.575)^3/12 + 0.1103(0.5585)^2 + 3.082(0.075)^3/12 + 0.2312(0.2665)^2 = 0.07372 \text{ in}^4$$

$$\lambda = 4.62(b/t) \sqrt{\frac{1 + A_s/bt}{1 + \sqrt{1 + (10.67 I_o)/bt^3}}} = 4.62(3.082/0.075) \sqrt{\frac{1 + 0.1103/0.075(3.082)}{1 + \sqrt{1 + 10.67(0.07372)/3.082(0.075)^3}}}$$

$$\lambda = 45.59 \quad \text{Eq. 3.4.9.2-6}$$

$$F_{bc} = \frac{27.64 - 0.1445(45.59)}{1.95} = 10.796 \text{ ksi}$$

$$F_{bt} = 20 \text{ ksi}$$

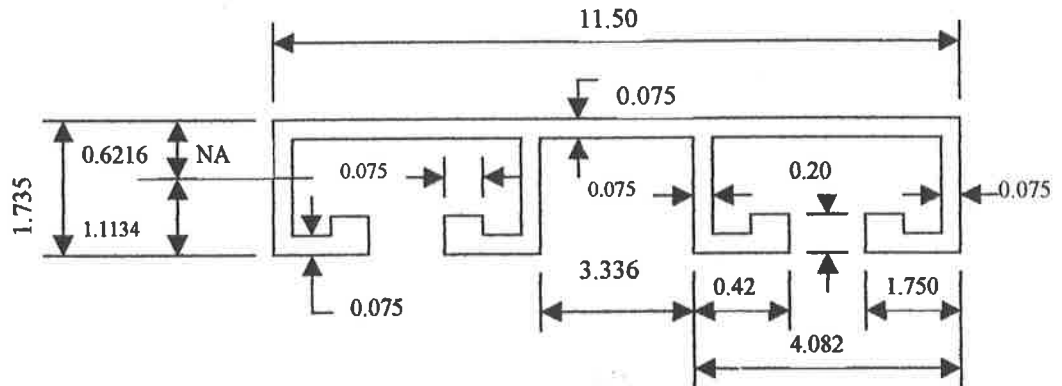
$$M = \delta S \quad \text{M-compression} = 10.796 (1.0826)/12 = 0.97 \text{ ft.-kips}$$

$$\text{M-tension} = 20 (0.5513)/12 = 0.92 \text{ ft.-kips}$$

$$\text{Moment (120 lbs./ln. ft)} = 120 \times 6^2/8 \times 1000 = 0.54 \text{ ft. kips} < 0.92 \text{ ft. kips.}$$

Plank is satisfactory for the maximum standard six (6) foot frame spacing.

LOAD CALCULATIONS FOR 2 x 12 ALUMINUM PLANK



$$\begin{aligned} \text{Area} &= 11.50 \times 0.075 = 0.8625 \text{ in}^2 \\ \text{Area} &= [2(1.750) + 2(0.420)](0.075) = 0.3255 \text{ in}^2 \\ \text{Area} &= 4(1.585)(0.075) = 0.4755 \text{ in}^2 \\ \text{Area} &= 4(0.075)(0.125) = 0.0375 \text{ in}^2 \\ \hline \text{Total Area} &= 1.7010 \text{ in}^2 \end{aligned}$$

$$\text{NA} = \frac{0.8625(1.6975) + 0.3255(0.0375) + 0.4755(0.8675) + 0.0375(0.1375)}{1.7010} = 1.1134 \text{ in}$$

$$\begin{aligned} I &= 11.50(0.075)^3/12 + 0.8625(0.5841)^2 + 4.34(0.075)^3/12 + 0.3255(1.0759)^2 + \\ &4(0.075)(1.585)^3/12 + 0.4755(0.2459)^2 + 4(0.075)(0.125)^3/12 + 0.0375(0.9759)^2 = 0.8357 \text{ in}^4 \end{aligned}$$

$$S_{x\text{-top}} = 0.8357/0.6216 = 1.3444 \text{ in}^3$$

$$S_{x\text{-bottom}} = 0.8357/1.1134 = 0.7506 \text{ in}^3$$

$$r_x = \sqrt{0.8357/1.7010} = 0.7009 \text{ in}$$

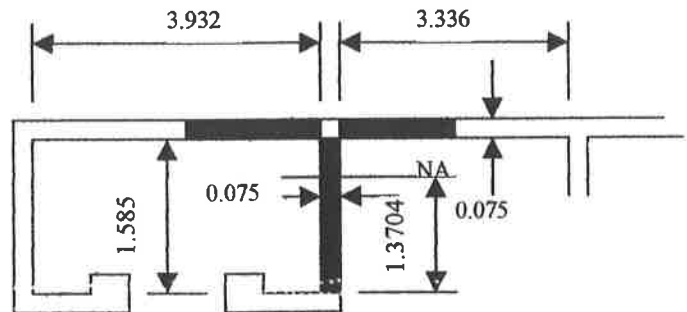
Reference: Allowable compressive stress in plank in accordance with Section 3.4.9.2 and allowable tension stress in accordance with Section 3.4.1 of the *Aluminum Design Manual*; The Aluminum Association 900 19th Street, N.W. Washington D.C. 20006 © 1994.

F_{cy} = Compressive Yield Strength = 25 ksi**
 B_c = Buckling Formula Constant - Compression
 D_c = Buckling Formula Constant - Compression
 $C_c = 0.41 \times B_c/D_c$ Table 3.3-4

$$B_c = F_{cy} \{ 1 + (F_{cy}/2250) \}^{1/2} \text{ Table 3.3-4}$$

$$B_c = 25 \{ 1 + (25/2250) \}^{1/2} = 27.64 \text{ ksi}$$

$$D_c = (B_c/10) \{ (B_c/E) \}^{1/2} \text{ Table 3.3-4}$$



$$D_c = 27.64/10 [(27.64/10100)/2] = 0.1445 \quad b = \frac{3.096 + 3.068}{2} = 3.082$$

$$C_c = 0.41 (27.64/0.1445) = 78.42$$

** F_c=25 ksi for 6063-T6 Table 3.3-1

$$S_1 = [B_c - (n_u F_{cy}/k_c n_y)] / D_c = [27.64 - (1.95 \times 25 / 1.12 \times 1.65)] / 0.1445 = 8.719 \quad \text{Eq. 3.4.9.2-4}$$

$$S_2 = C_c = 78.42 \quad \text{Eq. 3.4.9.2-5}$$

$$F_c = F_{cy} / k_c n_y = 25 / (1.12)(1.65) = 13.53 \quad \text{Eq. 3.4.9.2-1}$$

$$\text{Area stem} = 1.585(0.075) = 0.1189 \text{ in}^2$$

$$\text{Area flange} = \frac{3.634(0.075)}{0.3915} = 0.2726 \text{ in}^2$$

$$\text{Area Total} = 0.3915 \text{ in}^2$$

$$NA = \frac{0.1189(0.7925) + 0.2726(1.6225)}{0.3915} = 1.3704 \text{ in}$$

$$I_o = 0.075(1.585)^3/12 + 0.1189(0.5779)^2 + 3.634(0.075)^3/12 + 0.2726(0.2521)^2 = 0.0820 \text{ in}^4$$

$$\lambda = 4.62(b/t) \sqrt{\frac{1 + A_s/bt}{1 + \sqrt{1 + (10.67 I_o)/bt^3}}} = 4.62(3.634/0.075) \sqrt{\frac{1 + 0.1189/0.075(3.634)}{1 + \sqrt{1 + 10.67(0.0820)/3.634(0.075)^3}}}$$

$$\lambda = 53.75 \quad \text{Eq. 3.4.9.2-6}$$

$$F_{bc} = \frac{27.64 - 0.1445(53.75)}{1.65} = 12.044 \text{ ksi}$$

$$F_{bt} = 20 \text{ ksi}$$

$$M = \delta S \quad M\text{-compression} = 12.044 (1.3444)/12 = 1.35 \text{ ft.-kips}$$

$$M\text{-tension} = 20 (0.7506)/12 = 1.251 \text{ ft.-kips}$$

$$\text{Moment (120 lbs./in. ft)} = 100 \times 6^2/8 \times 1000 = 0.45 \text{ ft. kips} < 1.251 \text{ ft. kips.}$$

$$\text{Moment} = (200 \text{ lbs.})(6 \text{ ft}) / 4 \times 1000 = 0.30 \text{ ft. kips} < 1.251 \text{ ft. kips}$$

Plank is satisfactory for the maximum standard six (6) foot wide stair tread.

Bleacher Plank Clip Analysis

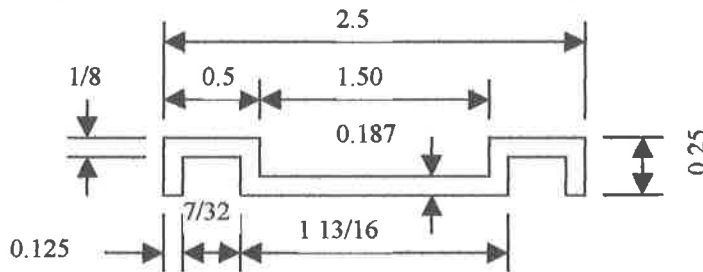
Check clip in bending for wind uplift and check the clip for shear using 5% of the live load or the side sway for the lateral load.

Wind Uplift 149 lbs. per plank attachment. 149 lbs./ 2 clips = 74.5 lbs. per clip uplift

Lateral load

24 lbs./ft. x 6 ft. = 144 lbs. 144 lbs. / 2 clips = 72 lbs. per clip ← Use this one.

720 lbs. x 0.05 = 36 lbs. 36 lbs./ 2 = 18 lbs. per clip



Minimum Side Shear Resistance

$$2 \times 5/16 \times 3/16 = 0.1172 \text{ in}^2$$

$$0.1172 \text{ in}^2 \times 8.5 \text{ ksi} = 996 \text{ lbs} > 72 \text{ lbs}$$

Bending Due to Wind Uplift

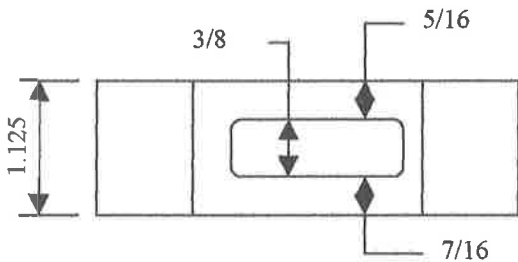
$$\text{Moment Arm} = l = \frac{1 \ 13/16 - 7/32}{2} =$$

$$l = 0.797 \text{ in}$$

$$M = 0.797 \times 74.5/2 = 29.7 \text{ in-lbs.}$$

$$S = bd^2/6 = 0.75 \times 0.1875^2 / 6 =$$

$$S = 0.0044 \text{ in}^3$$



$$F_b = 18.00 \text{ ksi} > f_b = 29.7 \text{ in-lbs} / 0.0044 \text{ in}^3 = 6750 \text{ psi}$$

The standard plank clip is adequate.

Aluminum alloy 6063-T6. Allowable stresses based on the *Aluminum Design Manual* published by the Aluminum Association.